

LTP

What does change relative velocity along the line of sight?

Rotation of the line of sight

Figure 4: The Firenze General Relativity experiment concept. Left: a portion from the Mission with one. Right: schematic of the frequency shift signal in a function of laser round trip time (B. Vitale et al., Nature, 428 (2004) p. 10).

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What does change relative velocity along the line of sight?

True forces that accelerate test-masses

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What does change relative velocity along the line of sight?

Interferometer measurement noise

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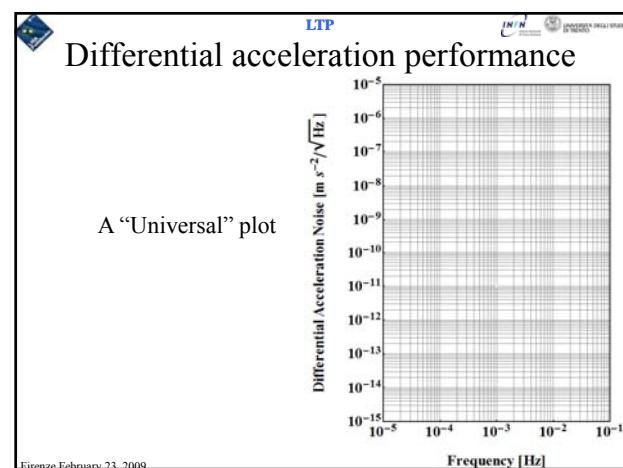
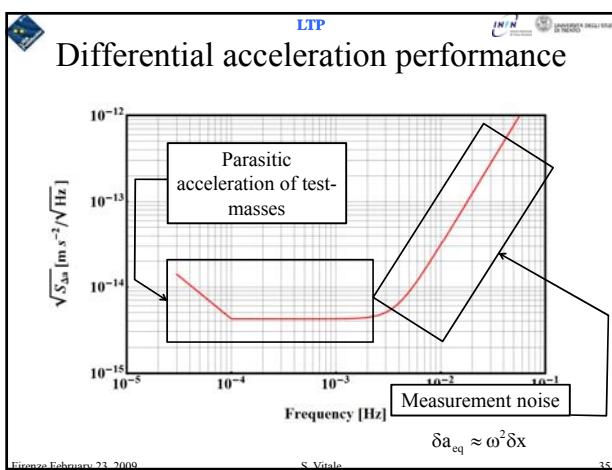
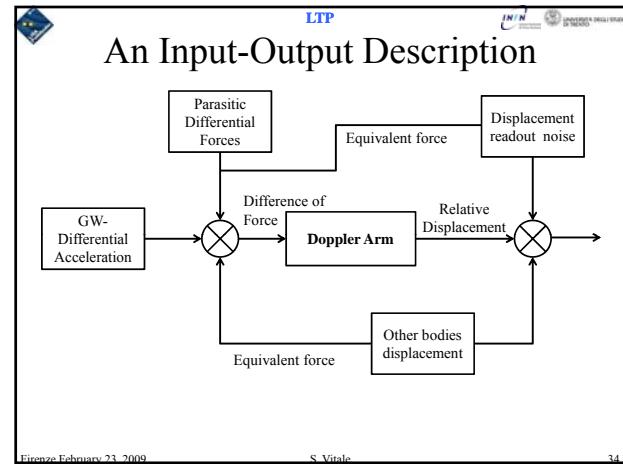
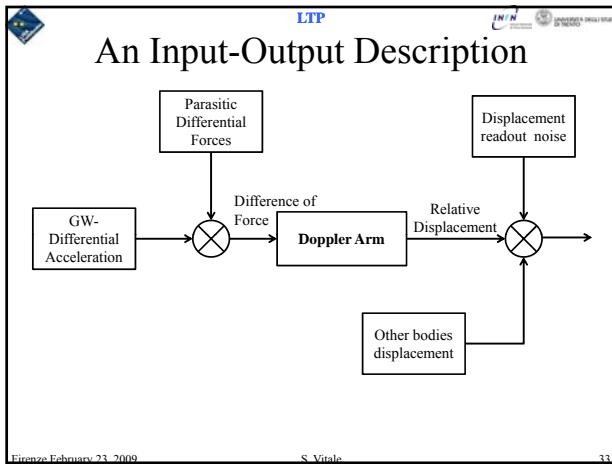
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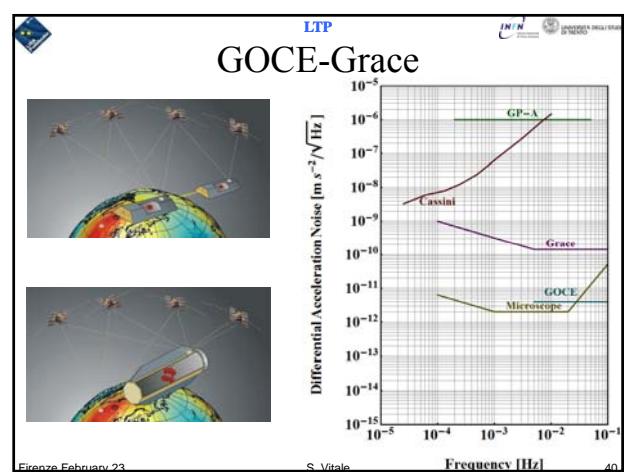
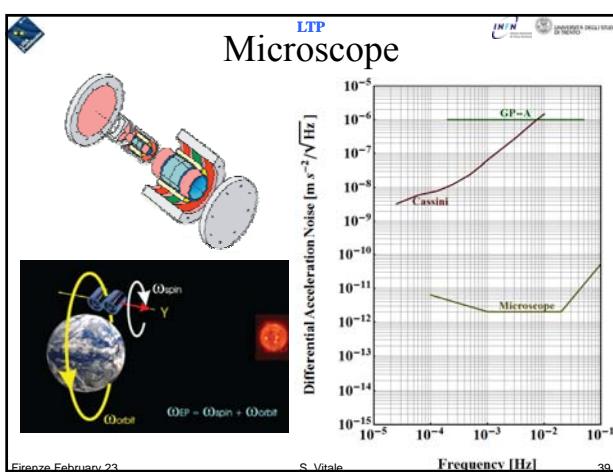
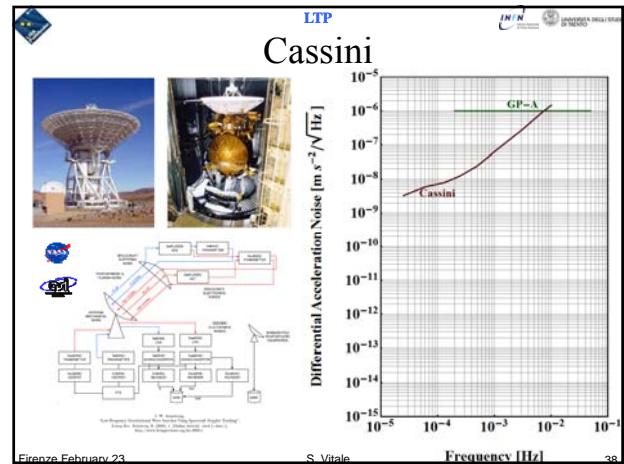
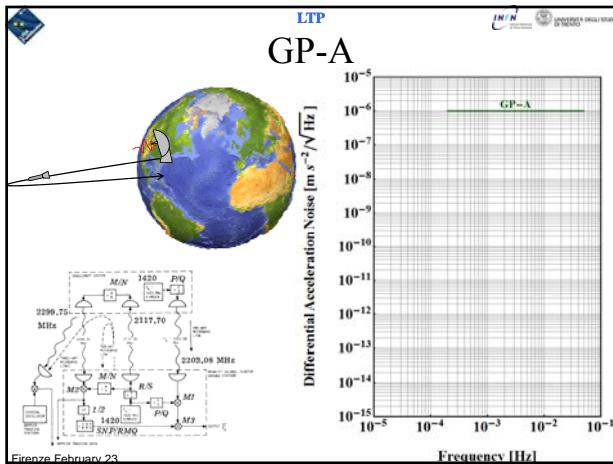
The problem of staged links

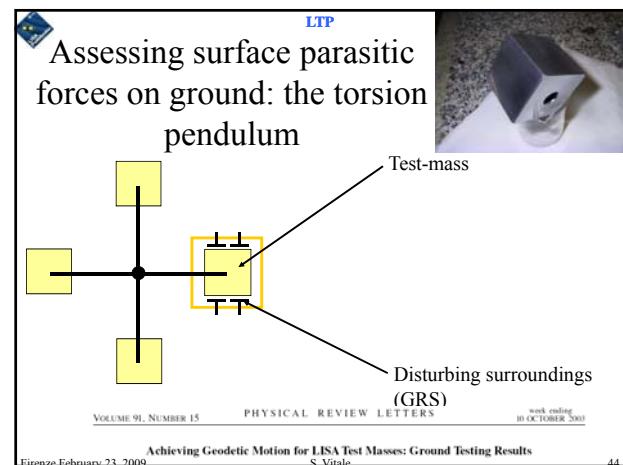
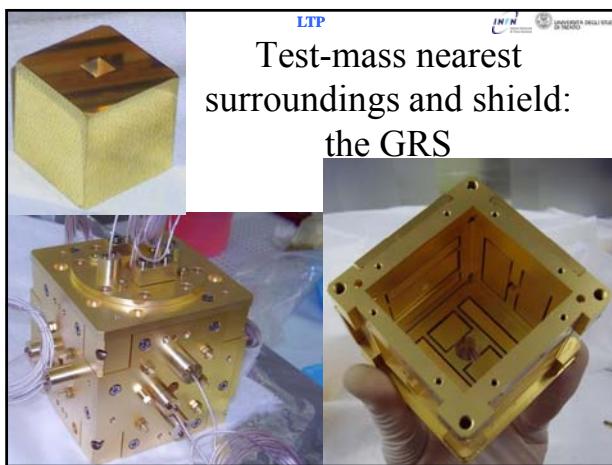
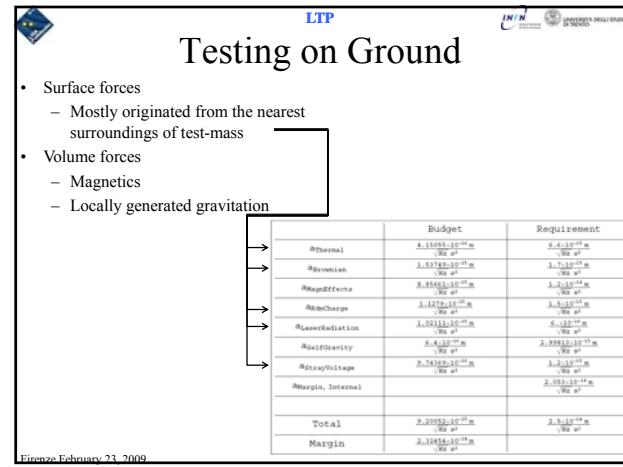
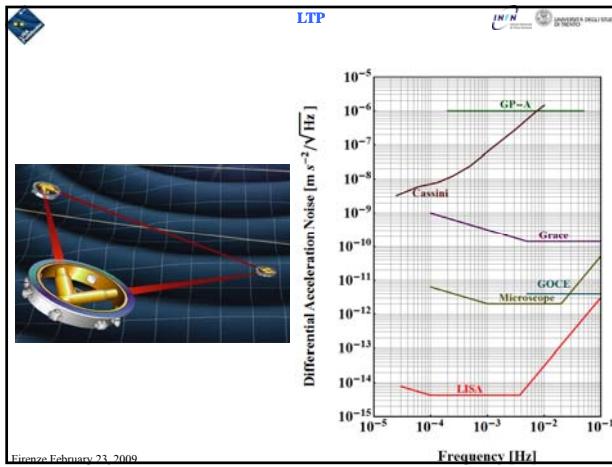
- Links are split as test-masses cannot carry optics
- Perfect split is insensitive to motion of body (bodies)
- Misalignments, calibration errors mix motion of extra bodies in

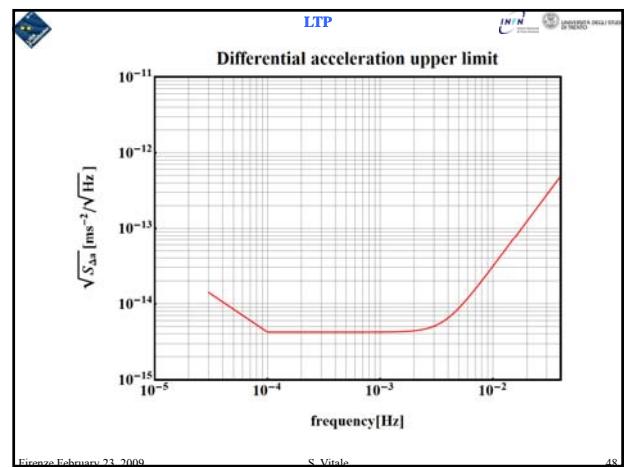
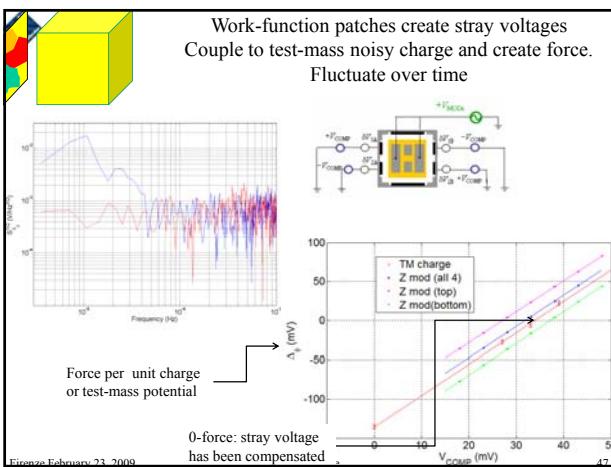
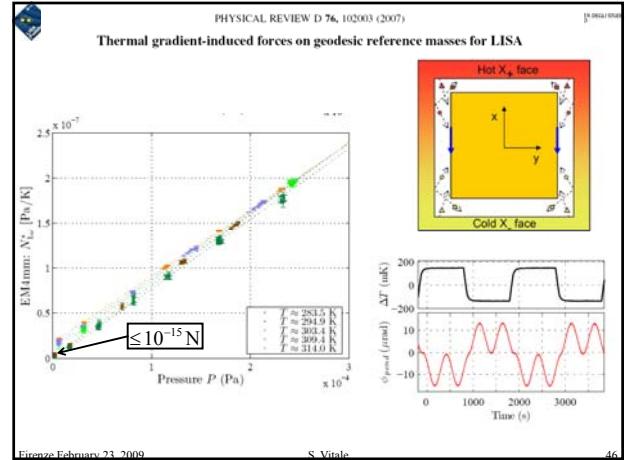
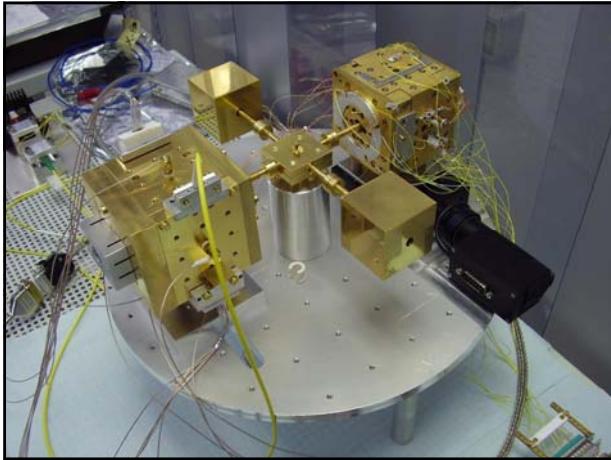
5 Mio. km

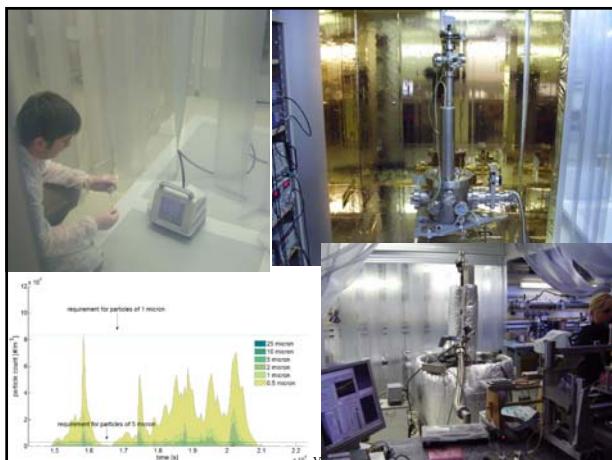
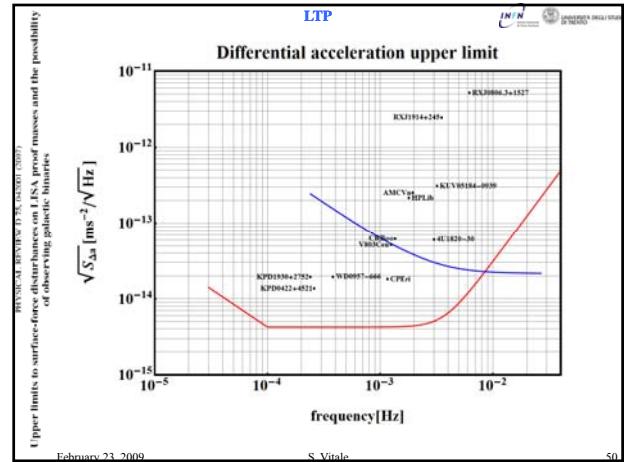
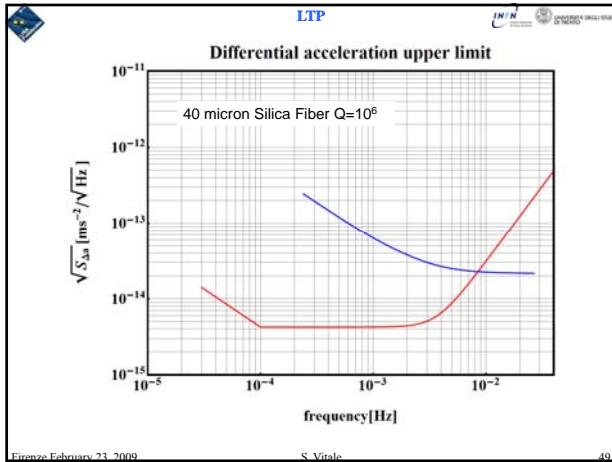
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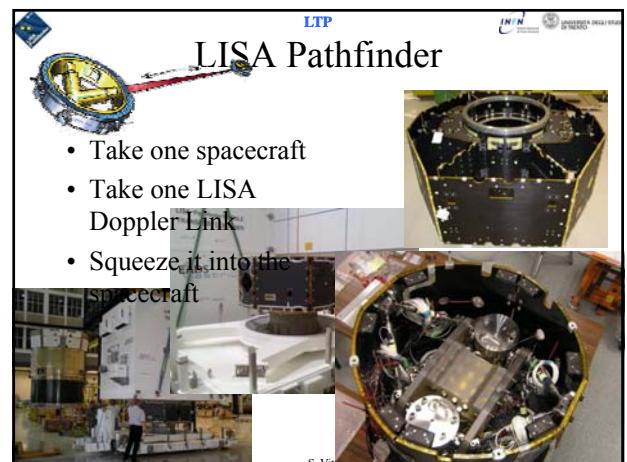
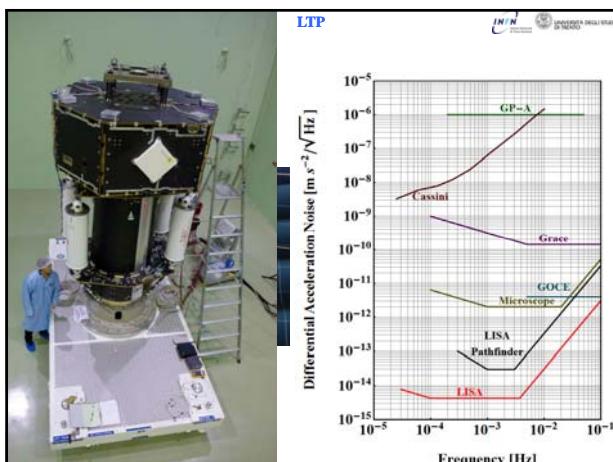
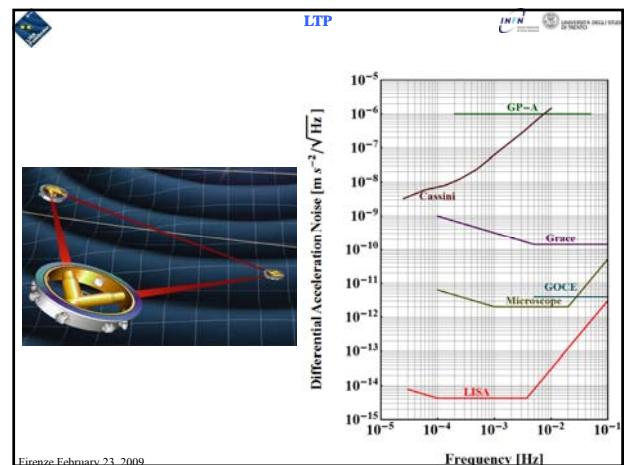


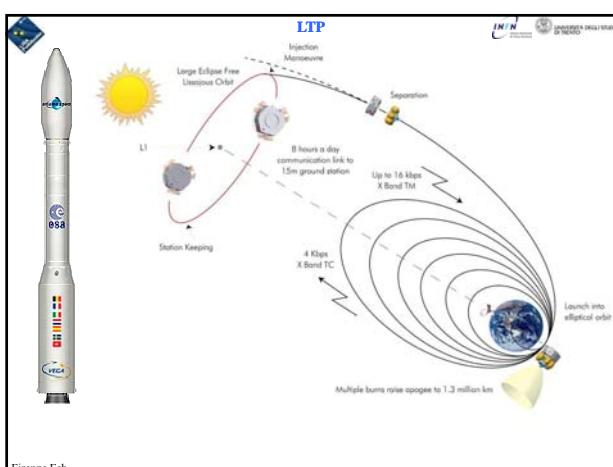
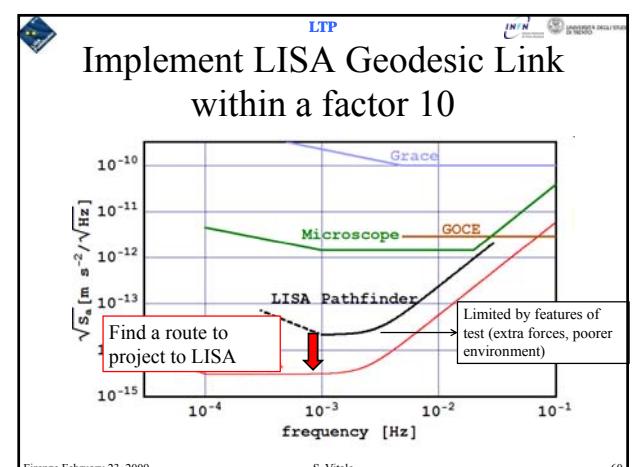
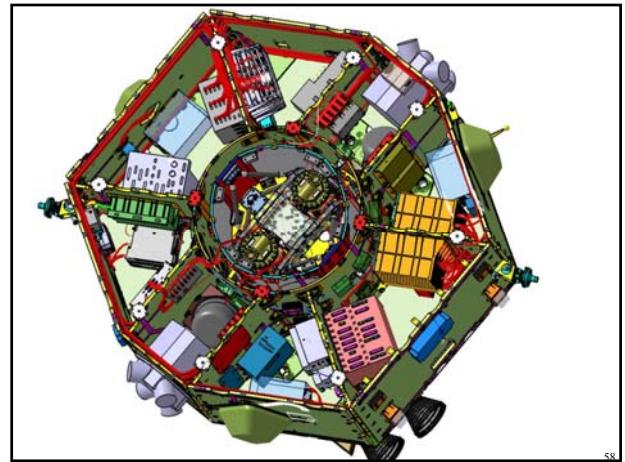
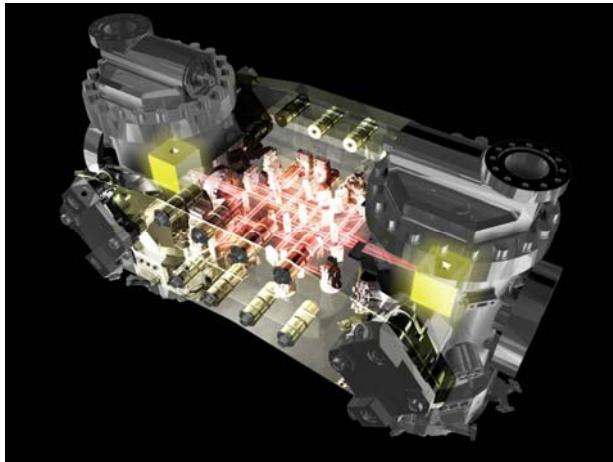










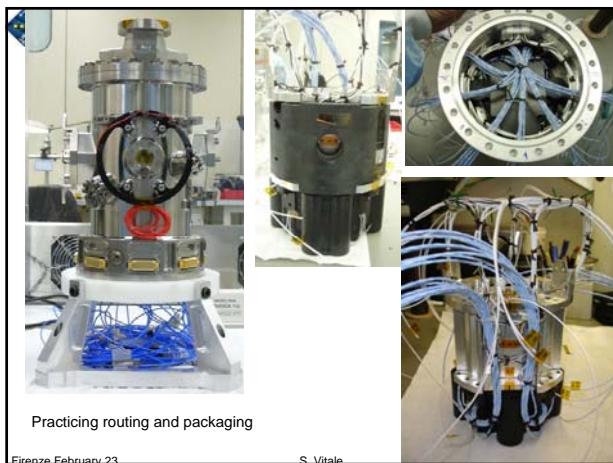
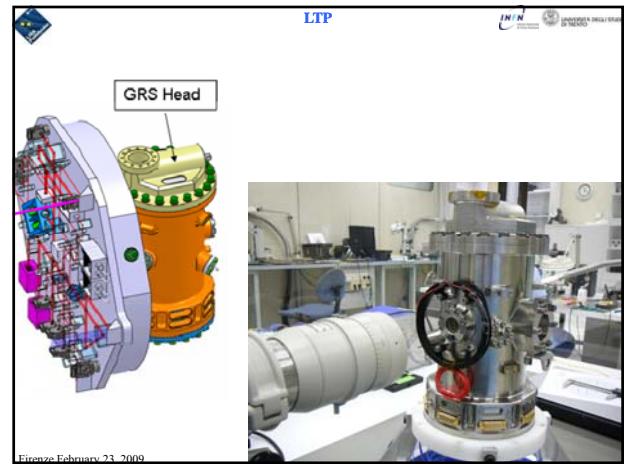


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Pathfinder → LISA

- Fly nominal LISA hardware on Pathfinder:
 - Maximize returns of the test
 - Shortens time to develop LISA
- Identify quantitatively leading sources of noise:
 - Physical model allows extrapolation to LISA
 - Will allow accurate understanding of LISA data

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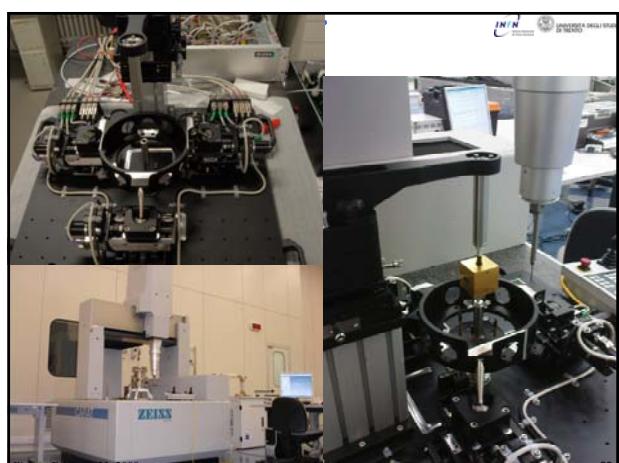
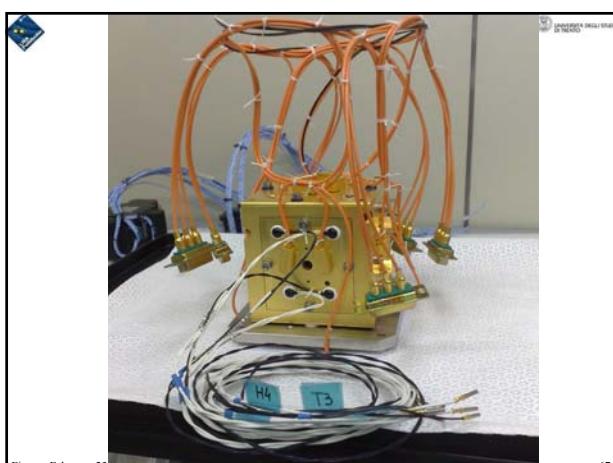
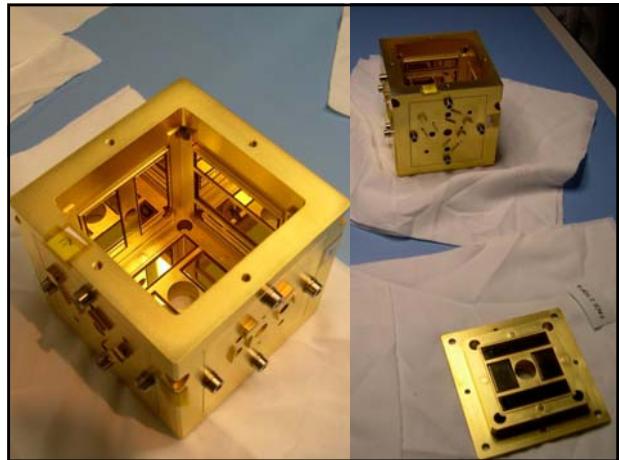
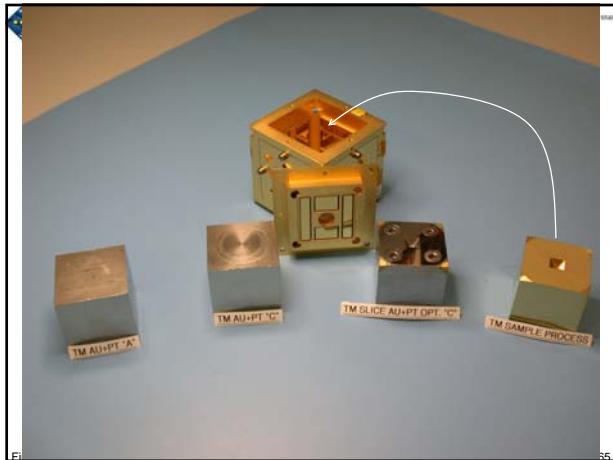
Gravitational balance

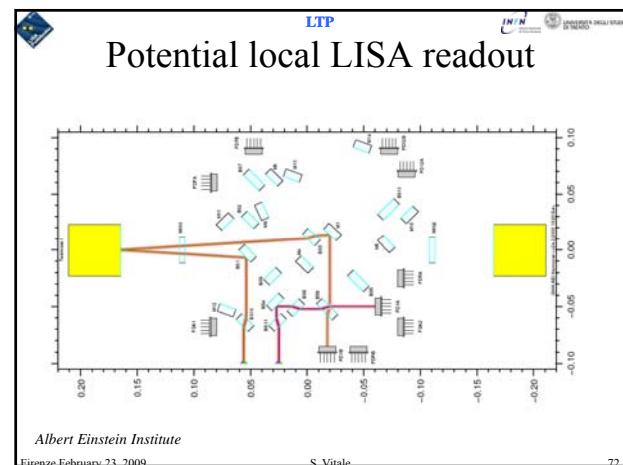
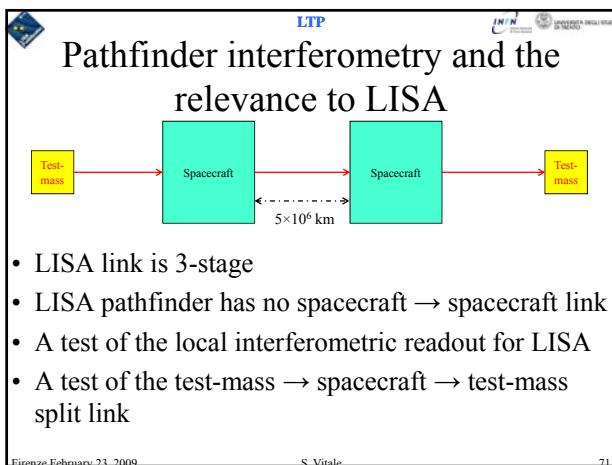
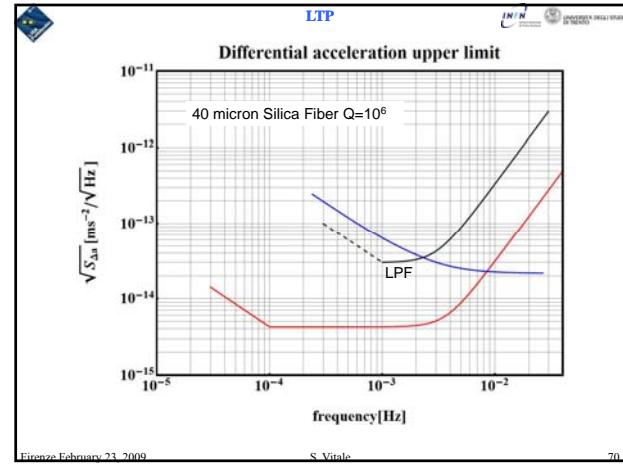
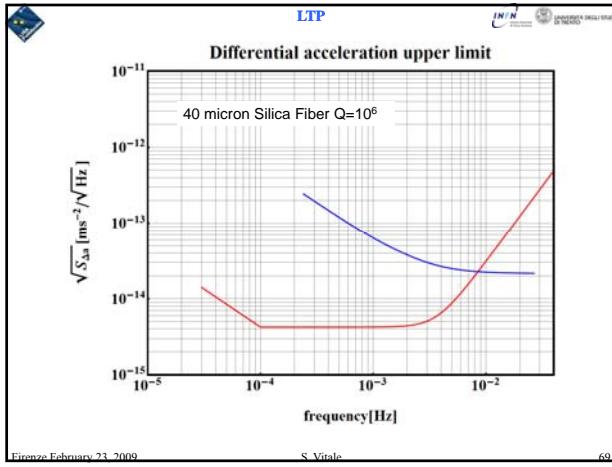
Residual (BM)	$\frac{g}{\sqrt{3}}$ on TM1 after compensation		
Mass (kg)	Cut along Z axis	Cut along Y axis	T _x
0.5511	-0.032	-0.046	0.377 -0.375 -0.377
0.5335	-0.03	-0.043	0.357 -0.363 -0.3448
0.5259	-0.034	-0.042	0.347 -0.353 -0.3448
0.5294	-0.042	-0.042	0.347 -0.345 -0.359
0.5295	-0.042	-0.042	0.347 -0.345 -0.359
0.5830	-0.04	-0.042	0.356 -0.359 -0.364
0.5831	-0.04	-0.042	0.356 -0.359 -0.364
0.54002	-0.04	-0.04	0.378 -0.376 -0.374
0.55501	-0.048	-0.04	0.347 -0.344 -0.368
0.54601	-0.048	-0.04	0.347 -0.344 -0.368
0.5487	-0.048	-0.04	0.356 -0.359 -0.364
0.54602	-0.04	-0.048	0.356 -0.359 -0.364
0.54603	-0.04	-0.048	0.356 -0.359 -0.364

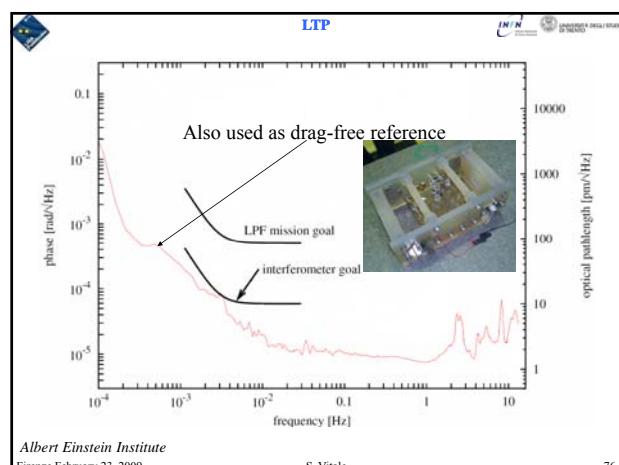
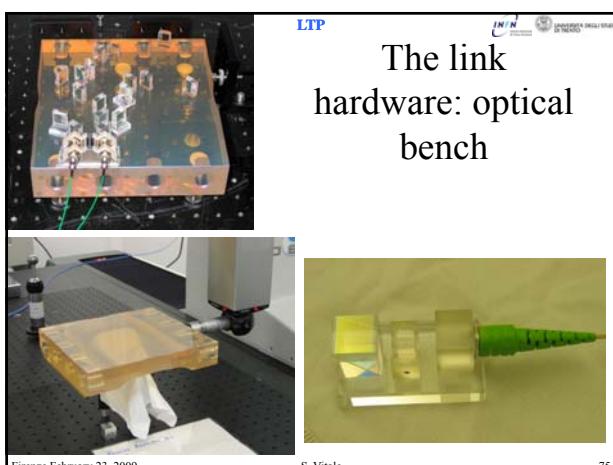
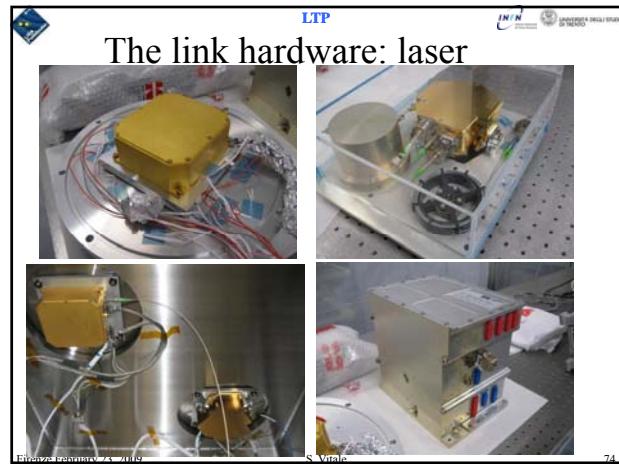
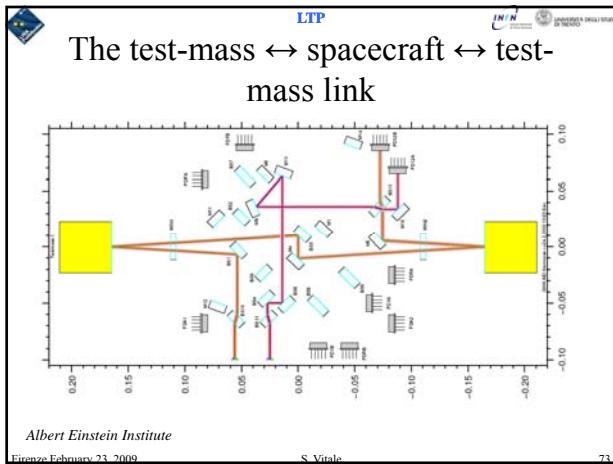
Table 22 Total Gravitational field (*italic characters* are the requirements)

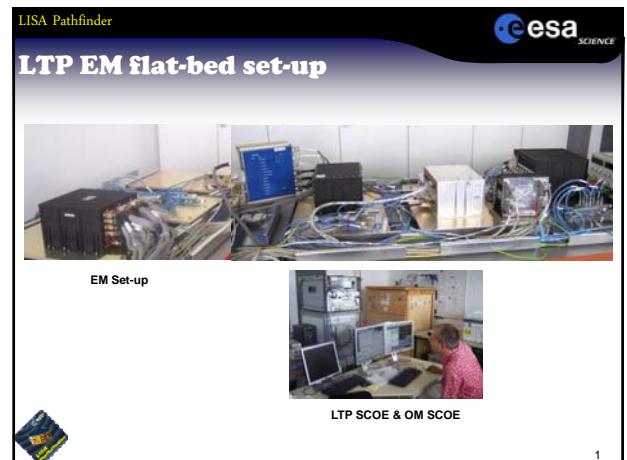
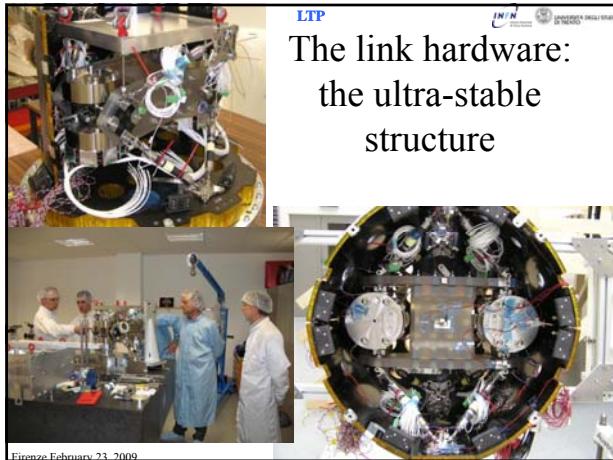
Figure 4-164 LCA Brackets Model Figure 4-54 Data Management Unit Model

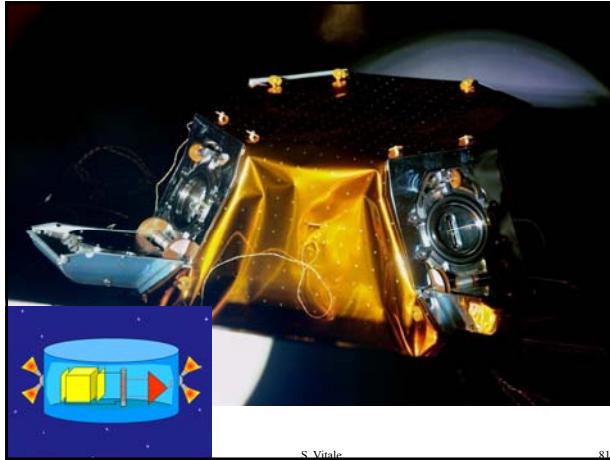
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Astrium	Experiment Performance Matrix	Experiment Performance Bu	Astrium	Experiment Performance Budget	LTP
Differential Rotation Pressure [DC#0200] ..	Magnetic Field Effects [R#1400]	Coriolis Error [R#1000]			
Gravitational Field [DC#0400] ..	(Under ASU responsibility)	(Under ASU responsibility)			
Margins [MEC#]	Magnetic Susceptibility	Total Software [R#2200]			
DCX Forces [DC#1000]	Magnetic field fluctuation [R#1400]	Differential Software along Sensative Axis [R#22100]			
Magnetic Field [DC#2000] ..	(Under ASU responsibility)	(Under ASU responsibility)			
(Under ASU responsibility)	Magnetic gradient fluctuation [R#1420]	Suffitess due to Electronic Actuation along the New			
Thermal Effects [DC#3000] ..	(Under ASU responsibility)	Terminae [R#22120] (Under ASU responsibility)			
Radiation Effect [DC#4000]	Dissipation of AC magnetic field [R#1430]	Absolute Suffitess on each Terminus along Sensative Axis [R#22300]			
Differential Radiation Pressure [DC#5000] ..	(Under ASU responsibility)	Suffitess due to Electronic Actuation along # [R#22320]			
Gravitational Wave [DC#6000] ..	Random velocity magnetic field fluctuations [R#15]	Suffitess due to Steady DC voltage effects [R#22340]			
Margins [MEC#]	(Under ASU responsibility)	Suffitess due to Thermal Effects [R#22360]			
Electric Field [DC#7000]	Low Frequency Noise [R#1440]	Suffitess due to Magnetic Field Effects [R#22380]			
Magnetic Field [DC#8000] ..	(Under ASU responsibility)	(Under ASU responsibility)			
(Under ASU responsibility)	Radiation Charging and Voltage Effects [R#1100]	Relative Positioning of the Field Effects [R#22390]			
Radiation Effect [DC#9000]	Random charge [R#1310]	(Under ASU responsibility)			
Differential Radiation Pressure [DC#1000] ..	Self Gravity Noise [R#1170]	Margins [M#22390]			
Gravitational Wave [DC#1100] ..	Self Gravity Noise from LTP [R#1170]	OMS Navigation [M#22400]			
Margins [MEC#]	Spatial noise [R#1170]	External Therm-Electric Induced Errors [R#41100]			
Thermal Effects [DC#1200] ..	Low Frequency Noise [R#11720]	Optical Manufacturing Measurement Noise [R#41200]			
Radiation Effect [R#1210]	(Under ASU responsibility)	Low Frequency [R#41210]			
Radiation Effect [DC#1300] ..	AC voltage down-conversion [R#11800]	Optical Pathlength Difference [R#41220]			
Asymmetric coupling [R#1310] ..	Margins [M#11800]	Position [R#41230]			
Thermal distortion of IR housing [R#1320] ..	Cross-Axes Coupling [R#13000]	Thermal Control Element Bench Tests [R#41250]			
It is the responsibility of the ASU	Cross-Coupling due to Suffitess [R#1100]	Slewlight [R#41260]			
Between Noise Terms [R#1400]	Part 5	Margins [M#1100]			
Distinct effects [R#1110]	Acceleration Performance Budgets	Part 6			
Resistor Back-action [R#1110]	Internal Forces [R#1000]	Background & Summary			
Unconnected part of resistor [R#1110]	Coriolis Effect [R#1110]	DC Optical Windshield [R#42100]			
Thermal Effects Within Sensor [R#1120]	Coriolis Effect [R#1110]	Thermal Effects [R#42110]			
Radiation Effect [R#1120]	Coupling of additive voltage noise at measurement	Thermal Effects [R#42120]			
Radiation Effect [R#1130] ..	Down conversion of additive voltage noise at AC conversion frequency (10MHz) [R#1400]	Thermal Effects [R#42130]			
Asymmetric coupling [R#1130] ..	[Hilbert Transform]	Low Frequency Effects [R#42140]			
Thermal distortion of IR housing [R#1132]	Cross-Axes Coupling [R#13000]	Angular Motion Effects [R#42140]			
It is the responsibility of the ASU	Cross-Coupling due to Suffitess [R#1100]	OMS-3 Internal Forces and Constraints [R#42200]			
Between Noise Terms [R#1132]	Part 6	Thermal Noise Forces [R#42300]			
Distinct effects [R#1132]	Background & Summary				
Resistor Back-action [R#1132]	DCX Forces [R#1000]	Thermal Effects [R#42400]			
Magnetic damping [R#1130] ..	Deep-Free Cross-Axis Control Error [R#1310]	Thermal Effects [R#42410]			
(Under ASU responsibility)	Suspension Cross-Axis Control Error [R#1310]	Thermal Effects [R#42420]			
Magnetic damping [R#1130]	Cross-Coupling due to Asymetric Armaton [R#13200]	Thermal Effects [R#42430]			
(Under ASU responsibility)	Coriolis Tilt Modes	Low Frequency Effects [R#42440]			
Magnetic damping [R#1130]	Suspension Cross-Tilt Modes	Angular Motion Effects [R#42450]			
(Under ASU responsibility)	Suspension Cross-Axes Cross-Tilt [R#13210]	Thermal Effects within Sensor [R#42460]			
(Under ASU responsibility)	Atmosphere Cross-Axes Cross-Tilt [R#13220]	Thermal Noise Terms [R#42470]			

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6.5 Mission goal: the physical model

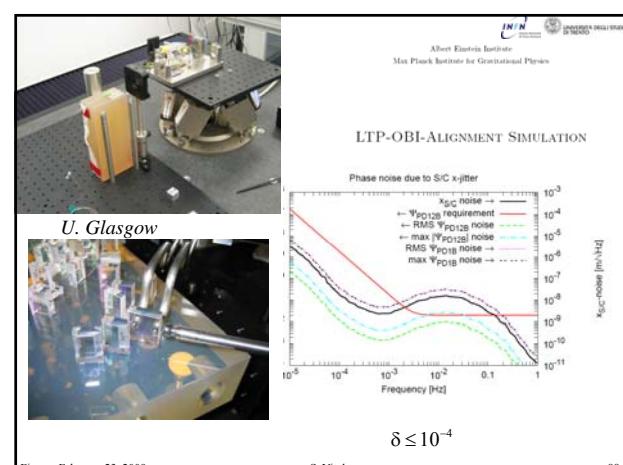
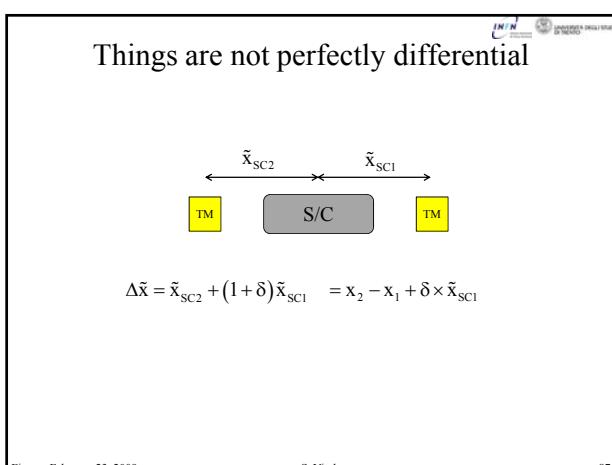
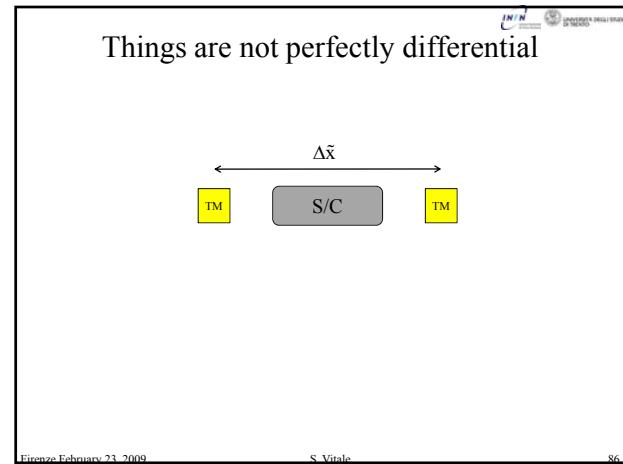
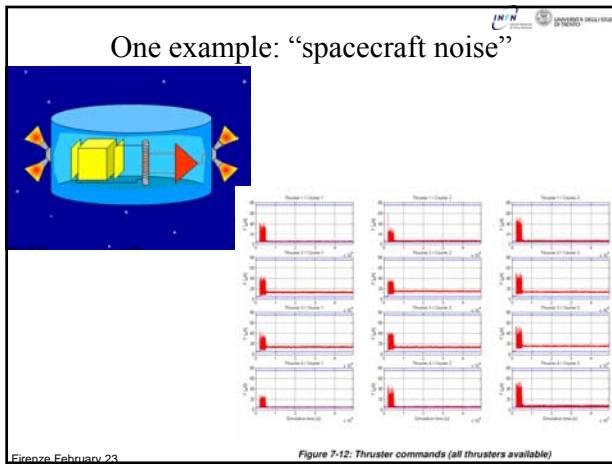
The final objective of LISA Pathfinder is to confirm the overall physical model of the forces that act on a test-mass in interplanetary space. To fulfil this program, the mission is not going to just make a measurement of acceleration but will implement a full menu of measurements:

- Measurement of acceleration noise between 0.0001 and 1 Hz.
- Measurement of dc-forces
- Measurement of force gradients
- Calibration of control loop transfer functions
- Characterization of thrust and thrust noise of micro-thrusters
- Measurement of interferometer performance and interferometer cross-talk
- Measurement of all cross-talk coefficients among different degrees of freedom
- Test of continuous charge measurement
- Test of continuous discharging and of discharging induced noise
- Test of magnetic induced noise
- Test of thermally induced acceleration noise
- Characterization of charging environment.

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The effect of gradients



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The effect of gradients



$$\mathbf{g}_1 = \omega_1^2 \tilde{\mathbf{x}}_{SC}$$

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The effect of gradients



Spacecraft jitter accelerates test-mass via gradients

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Correcting for Spacecraft Motion

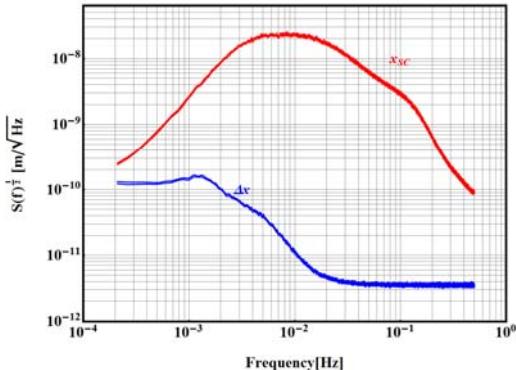
- Spacecraft motion is measured with high resolution (interferometer)
- Effect in differential link can be subtracted

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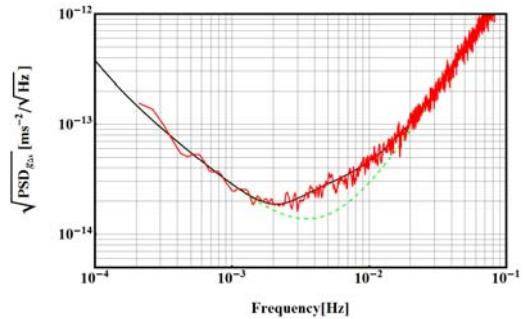
A simulated example



Firenze F.

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Uncorrected differential acceleration

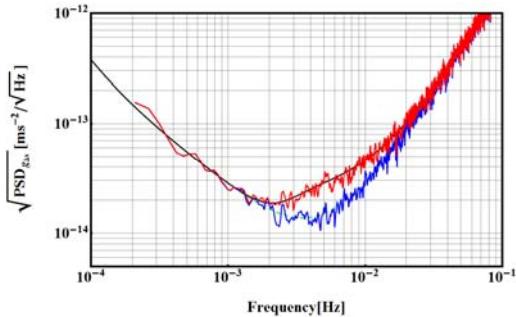


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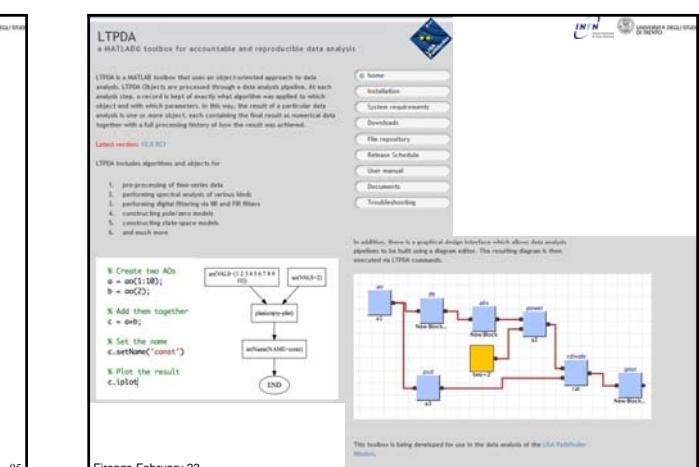
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Corrected for cross-talk and coupling



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